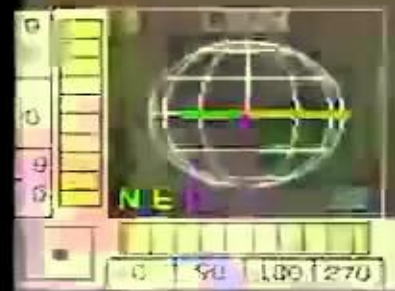


Time 294.08  
Altitude (Miles) 258.70  
Velocity (MPH) 11343.3



1  
00:00:49,310 --> 00:01:12,210  
meaning

2  
00:01:12,220 --> 00:01:21,510  
you

3  
00:01:26,190 --> 00:01:24,300  
and masti mlc on countdown one we've now

4  
00:02:05,820 --> 00:01:26,200  
complete our pre tech pre takeoff

5  
00:02:05,830 --> 00:02:10,480  
you

6  
00:02:10,490 --> 00:02:20,470  
now

7  
00:02:26,569 --> 00:02:23,600  
Elsie pegas go for launch I see it go

8  
00:02:31,490 --> 00:02:26,579  
for launch PLT confirm that's rough

9  
00:02:43,970 --> 00:02:31,500  
stand by for deployment TLT drop on my

10  
00:02:50,270 --> 00:02:43,980  
mark 321 drop this is away and ignition

11  
00:02:52,220 --> 00:02:50,280  
of the Pegasus rocket attitude remains

12  
00:02:58,369 --> 00:02:52,230  
nominal power buses remain strong with

13  
00:03:02,839 --> 00:02:58,379

good voltage on all buses this Pegasus

14

00:03:04,399 --> 00:03:02,849

is underway properly fins are following

15

00:03:12,470 --> 00:03:04,409

commands and controlling the flight of

16

00:03:14,569 --> 00:03:12,480

the vehicle attitude Ermine's nominal 35

17

00:03:20,960 --> 00:03:14,579

seconds into the iris mission we're past

18

00:03:23,149 --> 00:03:20,970

max Q vehicle power buses remain nominal

19

00:03:24,740 --> 00:03:23,159

vehicle is flying along the nominal

20

00:03:30,319 --> 00:03:24,750

predicted track indicating good

21

00:03:32,449 --> 00:03:30,329

performance at two demands nominal over

22

00:03:34,809 --> 00:03:32,459

50 seconds into the flight we're now

23

00:03:37,550 --> 00:03:34,819

over a hundred thousand feet in altitude

24

00:03:43,339 --> 00:03:37,560

traveling over 3000 correction four

25

00:03:45,920 --> 00:03:43,349

thousand miles an hour power buses

26  
00:03:47,629 --> 00:03:45,930  
remain strong approximately 10 seconds

27  
00:03:51,939 --> 00:03:47,639  
to stage one burnout in the stage one

28  
00:03:54,439 --> 00:03:51,949  
burn Stage one is burned out

29  
00:04:01,930 --> 00:03:54,449  
approximately 15 seconds to Stage one

30  
00:04:12,740 --> 00:04:08,210  
attitude remains nominal our bus is

31  
00:04:15,559 --> 00:04:12,750  
nominal stage one step in a stage

32  
00:04:21,520 --> 00:04:15,569  
separation stage 2 admission is

33  
00:04:26,170 --> 00:04:23,260  
our buses remain strong good voltage on

34  
00:04:28,060 --> 00:04:26,180  
all buses attitude remains nominal

35  
00:04:30,790 --> 00:04:28,070  
vehicle continues to fly along the

36  
00:04:32,980 --> 00:04:30,800  
nominal predicted track and at this

37  
00:04:34,990 --> 00:04:32,990  
point we lose our tracking of the chase

38  
00:04:38,140 --> 00:04:35,000

plane with the chase plane of the

39

00:04:40,750 --> 00:04:38,150

Pegasus will be looking forward with

40

00:04:42,310 --> 00:04:40,760

ground cameras if HTTP VC is operating

41

00:04:43,780 --> 00:04:42,320

nominally the nozzle is following

42

00:04:51,850 --> 00:04:43,790

commands and controlling the flight of

43

00:04:56,130 --> 00:04:51,860

the vehicle fairing separation confirmed

44

00:05:04,920 --> 00:04:58,420

attitude remains nominal after the

45

00:05:07,390 --> 00:05:04,930

deployment of the fairing approaching

46

00:05:09,940 --> 00:05:07,400

150 seconds into the iris mission all

47

00:05:12,100 --> 00:05:09,950

systems are performing nominally vehicle

48

00:05:17,670 --> 00:05:12,110

velocity is now eleven thousand miles an

49

00:05:25,150 --> 00:05:22,060

attitude remains nominal stage two is

50

00:05:27,550 --> 00:05:25,160

burned out vehicle is now in a prolonged

51  
00:05:33,190 --> 00:05:27,560  
Coast phase as we Coast up to the

52  
00:05:36,790 --> 00:05:33,200  
insertion altitude all systems are

53  
00:05:40,770 --> 00:05:36,800  
operating nominally now 180 seconds or

54  
00:05:44,920 --> 00:05:43,270  
telemetry is being broadcast from the

55  
00:05:48,279 --> 00:05:44,930  
stage three antennas and we have solid

56  
00:05:49,930 --> 00:05:48,289  
lock on telemetry in the center stage to

57  
00:05:54,610 --> 00:05:49,940  
predicted burnout point matches very

58  
00:05:56,200 --> 00:05:54,620  
nicely with the nominal prediction the

59  
00:05:58,150 --> 00:05:56,210  
flight computer on board the Pegasus

60  
00:06:00,400 --> 00:05:58,160  
vehicle has calculated the stage three

61  
00:06:04,600 --> 00:06:00,410  
ignition time switch the ignition time

62  
00:06:11,770 --> 00:06:04,610  
is 538 seconds and that is right on the

63  
00:06:13,330 --> 00:06:11,780

nominal value so the coast phase if the

64

00:06:16,060 --> 00:06:13,340

vehicle is in will last approximately

65

00:06:18,430 --> 00:06:16,070

five more minutes as we Coast up to an

66

00:06:25,810 --> 00:06:18,440

insertion altitude of approximately 328

67

00:06:37,010 --> 00:06:25,820

miles our buses remain strong telemetry

68

00:06:43,189 --> 00:06:40,490

now approximately 240 minutes into the

69

00:06:46,430 --> 00:06:43,199

mission the altitude is over 200 miles

70

00:06:48,529 --> 00:06:46,440

vehicle velocity is eleven thousand six

71

00:06:57,980 --> 00:06:48,539

hundred miles an hour all systems are

72

00:07:01,790 --> 00:06:57,990

operating nominally the nominal orbit

73

00:07:07,879 --> 00:07:01,800

for the iris spacecraft will be 380 7.5

74

00:07:09,830 --> 00:07:07,889

x 418 point seven miles all systems

75

00:07:12,200 --> 00:07:09,840

continue to operate nominally during the

76  
00:07:22,070 --> 00:07:12,210  
coast phase as the vehicle approaches

77  
00:07:28,159 --> 00:07:22,080  
the insertion altitude iris will orbit

78  
00:07:30,740 --> 00:07:28,169  
the earth every 97 minutes approaching

79  
00:07:32,959 --> 00:07:30,750  
300 seconds into the iris mission all

80  
00:07:35,779 --> 00:07:32,969  
systems are operating nominally the

81  
00:08:08,960 --> 00:07:35,789  
vehicle is over 250 miles in altitude at

82  
00:08:13,860 --> 00:08:11,640  
attitude remains nominal all power buses

83  
00:08:28,990 --> 00:08:13,870  
remain strong good voltage and expected

84  
00:08:33,909 --> 00:08:31,390  
we are approximately six minutes into

85  
00:08:36,490 --> 00:08:33,919  
the Pegasus iris mission all systems are

86  
00:08:38,740 --> 00:08:36,500  
operating nominally again the vehicle is

87  
00:08:42,580 --> 00:08:38,750  
still in the coast phase stage three

88  
00:08:44,740 --> 00:08:42,590

ignition is expected at 538 seconds and

89

00:09:01,590 --> 00:08:44,750

we are now at three hundred and sixty

90

00:09:06,540 --> 00:09:03,960

our buses remain nominal during this

91

00:09:12,620 --> 00:09:06,550

Coast phase the vehicle RCS system is

92

00:09:18,480 --> 00:09:15,569

now at four hundred seconds into the

93

00:09:27,269 --> 00:09:18,490

mission vehicles at an altitude of 329

94

00:09:30,650 --> 00:09:27,279

miles velocity is 1800 miles an hour our

95

00:09:50,530 --> 00:09:30,660

actual launch time was seven 2730

96

00:09:55,420 --> 00:09:52,930

on the countdown that we're now at 435

97

00:09:57,640 --> 00:09:55,430

seconds into the Pegasus iris mission

98

00:09:59,650 --> 00:09:57,650

all systems continue to perform

99

00:10:04,870 --> 00:09:59,660

nominally a little less than 100 seconds

100

00:10:07,210 --> 00:10:04,880

away from stage three ignition that's

101  
00:10:09,400 --> 00:10:07,220  
the voice of Steve Hollow the flight

102  
00:10:11,260 --> 00:10:09,410  
systems engineer approximately seconds

103  
00:10:13,030 --> 00:10:11,270  
the RCS system will be used to reorient

104  
00:10:23,110 --> 00:10:13,040  
the vehicle in preparation for stage 3

105  
00:10:29,170 --> 00:10:23,120  
ignition add a tutor mains nominal power

106  
00:10:31,630 --> 00:10:29,180  
buses remain strong vehicle is now

107  
00:10:34,180 --> 00:10:31,640  
maneuvering into the stage 3 ignition

108  
00:10:36,550 --> 00:10:34,190  
attitude RCS is performing nominally

109  
00:10:46,700 --> 00:10:36,560  
vehicle attitude remains nominal during

110  
00:10:54,050 --> 00:10:52,280  
stage 3 ignition and 30 seconds our

111  
00:10:56,510 --> 00:10:54,060  
buses remains strong vehicle

112  
00:10:58,190 --> 00:10:56,520  
reorientation maneuver is complete here

113  
00:11:07,130 --> 00:10:58,200

now with the appropriate stage 3

114

00:11:08,780 --> 00:11:07,140

ignition attitude approximately 20

115

00:11:10,850 --> 00:11:08,790

seconds will have stage to separation

116

00:11:25,420 --> 00:11:10,860

followed about 11 seconds later by stage

117

00:11:38,390 --> 00:11:30,680

stage to separation five seconds to

118

00:11:41,080 --> 00:11:38,400

stage three ignition stage three

119

00:11:43,520 --> 00:11:41,090

ignition attitude remains nominal

120

00:11:54,889 --> 00:11:43,530

telemetry beginning to take hits we are

121

00:11:59,569 --> 00:11:57,259

and this is peg on the countdown that we

122

00:12:01,249 --> 00:11:59,579

have LRS in the center all indications

123

00:12:04,160 --> 00:12:01,259

are that the vehicle is operating

124

00:12:09,079 --> 00:12:04,170

nominally when we went over the horizon

125

00:12:17,859 --> 00:12:09,089

for radio so stand by for iOS hopefully

126  
00:12:21,319 --> 00:12:17,869  
by airborne asset its loss of the

127  
00:12:29,389 --> 00:12:21,329  
tracking signal to Vandenberg stay by

128  
00:12:31,309 --> 00:12:29,399  
now for Dryden's dc-8 to acquire and we

129  
00:12:33,230 --> 00:12:31,319  
have a OS in the center we have good

130  
00:12:35,389 --> 00:12:33,240  
telemetry all systems appear to be

131  
00:12:40,040 --> 00:12:35,399  
continuing to operate nominally we're

132  
00:12:45,889 --> 00:12:40,050  
still in the stage through burn attitude

133  
00:12:49,879 --> 00:12:45,899  
remains nominal and we have stage 3

134  
00:12:52,160 --> 00:12:49,889  
burnout the vehicle is in orbit and we

135  
00:12:54,230 --> 00:12:52,170  
are at a nominal target orbit for the

136  
00:12:56,749 --> 00:12:54,240  
iris mission we're now going to enter a

137  
00:12:58,579 --> 00:12:56,759  
coast phase where we reorient the

138  
00:13:01,069 --> 00:12:58,589

vehicle and preparation for payload

139

00:13:03,769 --> 00:13:01,079

separation that will last approximately

140

00:13:08,239 --> 00:13:03,779

two minutes and 30 seconds from this

141

00:13:10,639 --> 00:13:08,249

point forward attitude remains nominal

142

00:13:13,129 --> 00:13:10,649

the vehicle RCS is being used to control

143

00:13:15,019 --> 00:13:13,139

the vehicle attitude and is now pointed

144

00:13:23,629 --> 00:13:15,029

in the appropriate direction for iris

145

00:13:25,759 --> 00:13:23,639

payload separation our buses remain

146

00:13:28,970 --> 00:13:25,769

strong good voltage and good curl on all

147

00:13:33,290 --> 00:13:28,980

buses yes Deon haven't go ahead how we

148

00:13:35,569 --> 00:13:33,300

do attitude remains nominal where 650

149

00:13:43,730 --> 00:13:35,579

seconds into the mission expect a load

150

00:13:45,769 --> 00:13:43,740

separation at 786 seconds copy that and

151  
00:13:54,300 --> 00:13:45,779  
GNC reports that we have an excellent on

152  
00:13:58,500 --> 00:13:56,250  
this peg in the center we are

153  
00:14:09,320 --> 00:13:58,510  
experiencing some loss of telemetry

154  
00:14:12,000 --> 00:14:09,330  
which is just due to link margin and

155  
00:14:29,840 --> 00:14:12,010  
this is pegging the center we have elbow

156  
00:14:35,680 --> 00:14:32,449  
and this is peg on the count on that we

157  
00:14:37,999 --> 00:14:35,690  
do have lrs by local vandeberg sites

158  
00:14:41,420 --> 00:14:38,009  
unfortunately we do not have the

159  
00:14:43,490 --> 00:14:41,430  
airborne asset collecting telemetry so

160  
00:14:45,050 --> 00:14:43,500  
that is the reason that we hire LS it is

161  
00:14:57,620 --> 00:14:45,060  
not associated with the performance of

162  
00:14:59,180 --> 00:14:57,630  
the Pegasus launch vehicle this is peg

163  
00:15:01,129 --> 00:14:59,190

on the countdown that we still have el

164

00:15:03,019 --> 00:15:01,139

oso reason for that is that the airborne

165

00:15:05,900 --> 00:15:03,029

aircraft that was sent up to receive

166

00:15:07,280 --> 00:15:05,910

telemetry downrange did not pick up a

167

00:15:09,590 --> 00:15:07,290

signal that has not been able to track

168

00:15:10,670 --> 00:15:09,600

the signal that was not a problem

169

00:15:13,639 --> 00:15:10,680

associated with the Pegasus launch

170

00:15:15,350 --> 00:15:13,649

vehicle as we saw the local range

171

00:15:27,190 --> 00:15:15,360

telemetry was strong until we went over

172

00:15:27,200 --> 00:15:33,180

payload sep would occur in 12 seconds

173

00:15:48,010 --> 00:15:39,120

and just for note the UTC of drop was 02

174

00:15:49,510 --> 00:15:48,020

2746 UTC of drop was 02 2746 predicted

175

00:15:50,980 --> 00:15:49,520

payload separation and this is just

176  
00:15:54,250 --> 00:15:50,990  
based on the timer running here in the

177  
00:15:56,110 --> 00:15:54,260  
center is now we do not have telemetry

178  
00:15:58,000 --> 00:15:56,120  
to confirm that however we do have

179  
00:16:00,250 --> 00:15:58,010  
enough data to provide the spacecraft

180  
00:16:08,710 --> 00:16:00,260  
with a state vector so that is in work

181  
00:16:10,900 --> 00:16:08,720  
at this time this is peg on countdown

182  
00:16:13,960 --> 00:16:10,910  
we'll go ahead and report the orbit that

183  
00:16:17,350 --> 00:16:13,970  
we had at the loss of telemetry we add

184  
00:16:19,510 --> 00:16:17,360  
an apogee of six hundred and sixty nine

185  
00:16:23,460 --> 00:16:19,520  
point nine eight kilometers which is

186  
00:16:26,770 --> 00:16:23,470  
right on target at a perigee of 622 dot

187  
00:16:32,560 --> 00:16:26,780  
96 again which is right on target an

188  
00:16:37,720 --> 00:16:32,570

inclination of 97 dot 899 again on

189

00:16:39,490 --> 00:16:37,730

target and an ml tan of 0 60 235 again

190

00:16:42,190 --> 00:16:39,500

right on target so a great great

191

00:16:49,000 --> 00:16:42,200

insertion orbit for iris state vector is

192

00:16:55,660 --> 00:16:49,010

in work an update to our launch time 620

193

00:16:58,210 --> 00:16:55,670

or other 727 44 decimal 0007 2744

194

00:17:00,820 --> 00:16:58,220

decimal zero zero zero and that we have

195

00:17:03,370 --> 00:17:00,830

a report that iris is being tracked by

196

00:17:05,740 --> 00:17:03,380

the teachers relay satellites those

197

00:17:07,270 --> 00:17:05,750

satellites are reading telemetry from

198

00:17:09,760 --> 00:17:07,280

the spacecraft which is obviously

199

00:17:11,680 --> 00:17:09,770

separated successfully from Pegasus to

200

00:17:14,230 --> 00:17:11,690

their ground stations so congratulations

